

Article

# Instantaneous Rotational Speed Algorithm for Locating Malfunctions in Marine Diesel Engines

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Received: 18 February 2020; Accepted: 15 March 2020; Published: 17 March 2020

**Abstract:** This article suggested broadening the standard methods for diagnosing the technical condition of diesel engines to include an analysis of the instantaneous rotational speed of compression combustion engines with the use of a novel algorithm. The authors revised the subject concerning the use of the analysis of instantaneous changes in the rotational speed of an engine when assessing its technical condition and the location of the malfunction. An algorithm and its practical implementation in a prototype diagnostic system called SpeedMA were presented. This article reported the test results of the prototype in the context of indicating the engine cylinder in which ignition failed to occur. Tests were carried out for two marine engines: a low-speed trunk engine directly driving the fixed-pitch propeller and a medium-speed trunk engine driving the alternator. For each case, an analysis was carried out for different engine loads and at individual cylinders in which combustion failed to occur. The experimental results showed an unambiguous relation between the combustion process of the examined engines and changes in the instantaneous rotational speed. The results also confirmed the usefulness of the proposed method and showed the correct operation of the presented diagnostic algorithm. The proposed diagnostic system could be used during the operation of engines running in real ship engine rooms.

**Keywords:** engine speed measurements; marine engine; combustion process; cylinder misfire; condition monitoring; engine diagnostics, load balance analysis, speed uniformity analysis

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## 1. Introduction

The combustion process occurring in the combustion chambers of a marine reciprocating engine and its analysis are the main sources of information involved in the operation and condition of the components of a running engine. This provides a range of information on the condition of the injection device, the piston-crank system, the valvetrain, and the quality of the burning fuel [1–4]. One of the most common malfunctions found in marine engines is damage to the fuel system, which directly affects the instantaneous rotational speed [5–7]. High-powered compression-ignition marine diesel engines are used as propulsion systems and energy sources for ship power plants and are equipped with engine supervision systems that utilize sensors and advanced mathematical methods [8–12]. Remote and local exhaust gas temperature sensors are the basic diagnostic and monitoring tools. The fuel combustion process involves temperature fluctuations of the exhaust gases, depending on the load and the quality of combustion itself [13–16]. Depending on its source, most malfunctions during the operation of injection devices either increase or decrease the exhaust gas temperature at the cylinder exit.

More advanced monitoring methods involve systems based on sensors that measure the combustion pressures in combustion chambers [17–20]. For example, this concept has been used by ABB (ASEA Brown Boveri corporation) in their Cylmate product [18] using a system that consists of pressure sensors equipped with piezoelectric crystals that deform under pressure mounted on each cylinder and a shaft position transmitter mounted on the engine flywheel. Each component is

connected to the Cylmate transducer busbar, which collects real-time data from each motorcycle via the CAN (Controller Area Network) bus. A built-in mathematical model of the engine calculates the position of the crankshaft in order to obtain the correct angle of the piston position at each cylinder. The most important combustion parameters, such as the maximum combustion pressure, angle of maximum combustion pressure, scavenging pressure, compression pressure, and average pressure, are recorded and monitored for each cycle and can be displayed in the form of charts [21–24]. All deviations from normal values are indicated as an alarm. Evaluated data, alarms, and events are transmitted via Ethernet LAN to the Cylmate control unit and also to higher-level systems that monitor the entire engine room. This device is excellent for monitoring the combustion process in low-speed engines. Periodic clogging of the sensor inlet channels and the service life of the sensors themselves are considered to be operating defects.

Strain gauge sensors mounted on shafts or encoders monitoring the degree of torsion of the drive shaft are treated as a separate group of monitoring equipment [25–27]. KYMA KPM [26] is an example of the first type of system—a shaft power meter that measures torque, resistance generated by the drive, revolutions, and power output of the motor by continuously measuring the current flowing through a strain gauge attached to the drive shaft. Devices based on the measurement of shaft torsion using a mounted at a precise distance on the shaft are an alternative to strain gauges. The measurement of pulses from encoders and their mutual shifting in time inform the algorithm about the degree of shaft torsion. These devices, such as the LEMAG Shaftpower, improve operating efficiency, protect against overloads, and help optimize fuel consumption and save energy [28,29].

The angular velocity of an engine is not constant and changes momentarily, depending on the forces/moments acting on the crank-piston system [30–33].

An engine running under constant load produces an effective power equal to [1]:

$$N_e = \frac{M_o \alpha_n}{1000 \cdot \tau} = \frac{M_o \omega}{1000} = \frac{M_o 2\pi r}{60 \cdot 1000} = \frac{M_o n}{9,549296 \cdot 10^3} \quad (1)$$

where:  $\omega$  (rad/s)—the angular speed of the crankshaft,  $\alpha_n$  (rad)—crank angle,  $\tau$  (s)—time.

Transforming the above formulae produces a proportional dependence of the rotational speed of the engine  $n$  (rpm) on the effective power  $N_e$  (kW), and an inversely proportional dependence of the torque  $M_o$  (Nm) acting on the engine is observed:

$$n \approx 9.55 \cdot 10^3 \frac{N_e}{M_o} \quad (2)$$

Analyzing the dynamic torque equation of the internal combustion engine crankshaft [30,34], shown in Formula 3, confirms the correlation between the change in angular velocity and, after transformation, the correlation of the instantaneous rotational velocity for a given moment, and the torques acting on a given engine cylinder system. Because the torque during engine operation constantly changes, the instantaneous speed also changes:

$$J \frac{d\omega}{dt} = J(\varphi) \frac{d^2\varphi}{dt^2} = M_T - M_{B\alpha} - M_{ob} - M_t \quad (3)$$

where:  $J$  (kg m<sup>2</sup>)—reduced inertia mass moment of crank-piston system components,  $\varphi$  (rad)—angular position,  $M_T$  (Nm)—tangential force moment  $T$ ,  $M_{B\alpha}$  (Nm)—moment of inertia forces,  $M_{ob}$  (Nm)—load moment,  $M_t$  (Nm)—friction and pumping moment.

## 2. State-of-the-Art

The change in the instantaneous rotational speeds recorded over the course of the research by the authors was, to a large extent, consistent with the course of the combustion process occurring in the engine and was in-line with other studies [35–37]. Modern science is engaged in advanced research on measuring the speed of piston engine crankshafts and its use in diagnostics and operation [38–40].

According to the best knowledge of the authors, the papers tackling the issue of the engine speed uniformity analysis mostly discuss detecting the angular speed deviation changes. The most

important works are presented in Table 1. Among others, Biocanin, Dereszewski, Xiang, and others have published a number of papers, demonstrating the link between the instantaneous angular velocity and the proper course of combustion or malfunction detection. Dereszewski reported the calculation of moments acting on the piston-crank system by simultaneously testing the angular velocities using two encoders [41]. In turn, in his dissertation, Johnson presented a speed deviation calculation model that considered implemented neural networks [42]. Charchalis demonstrated and presented equations describing how the relationship between speed and torque affected a rotating crankshaft [30]. Wang et al. proposed using image correlation and periodicity algorithms [43]. Desbazeille and Zhang extended the analysis to V engines [44,45]. In his article, Dauglas attempted to combine the analysis of instantaneous rotational speed with vibroacoustic methods [46]. Also, research of other than diesel engines machines based on instantaneous rotational speed has been conducted [47,48].

Apart from the mentioned literature on speed deviation analyses, there is very little information on the algorithms suitable for use in online equipment monitoring the engine condition. The paper discussed the *W* coefficient calculation method, which covered deviation in angular speed for real engine. *W* coefficient was calculated for several engines with a different purpose (propulsion engine and electric power plant) and a different response from the engine governor (Woodward UG-8 type governor). All measurements showed that our method/algorithm was correct and could be used as a logic source for future use in the detector. The first prototype has already been created, and it detects the missing fire in units.

There are a number of engines without indicating cocks, and these engines are equipped only with basic diagnostic systems, e.g., cooling water temperature measurement and exhaust gas manifold temperatures. Moreover, these engines have a very compact construction, which makes them difficult to diagnose with on-line systems and with workshop parts measurements and verification. Furthermore, the second approach is time-consuming. Most of the marine engines installed on offshore support vessels, manufactured by, e.g., Detroit Diesel, John Deere, Caterpillar, Daimler Chrysler, Scania, MAN, Volvo, etc., might be classified to this group. The system proposed by the authors might be a good alternative for such engines as support for marine engineers. It might reduce the costs of operation and downtime periods.

**Table 1.** Comparison between the proposed instantaneous speed monitoring system and the works of other researchers.

Authors/System	Engine Maker	Nominal/ Test Speed (rpm)	Type	Cylinder Number	Nominal/ Test Power (kW)	Monitored System Type	Application	Data Presentation /Type of Diagram
SpeedMA System [49,50]	Sulzer 5 BAH 22	500	L	5	220	Marine gen-set	Load balance evaluation	Orthogonal, polar
	Buckau-Wolf R8DV-136	180-360	L	8	220	Marine main engine	, cylinder misfire detection, and location	
S. Biočanin, M. Biočanin [51]	OM403	2200	V	10	287	Vehicle engine	Misfire detection and location	Orthogonal
P. Charles et. al. [52]	No data	740, 1000	V	16, 20	8400, 9000	No data	Engine failure detection	Polar
M. Dereszewski, A. Charchalis [30,36]	Sulzer 3BAH22	600–750	L	3	100–220	Marine gen-set	Engine failure detection	Orthogonal
M. Desbazeille et. al. [44]	Wartsila diesel V50*	1500	V	20	4000	Marine/stationary gen-set	Faulty fuel pump detection	Orthogonal
Y. He et. al. [53]	Weichai WP10-240	21,002,200	L	6	175	Vehicle engine	Misfire detection	Orthogonal

R. Johnsson [38,42]	Scania DCS9 08	800–2200	L	6	280	Test engine modified for the experiment	Prediction of the maximum combustion pressure	Orthogonal
Z. Li et al. [54]	Volvo Penta TAMD165 C	1500	L	6	ca. 560	Marine gen-set	Engine fault diagnosis	Orthogonal, spectral
I. Margaronis [55]	Mirrlees JVSS12	750	No data	6	920	Marine main engine	Analysis of torque and cylinder pressure correlation	Quantitative measures
G. Rizzoni [37]	No data	No data	V	6	N/A	Vehicle engine	Cylinder misfire detection and location	Quantitative measures
F. Tagliatela et al. [56]	PFI	2500	L	1	10	Test gasoline engine	Maximum combustion pressure estimation	Quantitative measures
J. Yang et al. [57]	4120 SG	800–1400	L	4	48.5	Test stand	Engine failure detection	Quantitative measures

A literature analysis showed that there are currently no widely applied diagnostic systems using rotational speed measurements to evaluate the combustion process in marine systems. Occasional use of instantaneous velocity in prototype diagnostic systems is not supported by validated data processing algorithms to diagnose an engine condition or locating malfunctions during its operation. The authors proposed an innovative algorithm for calculating deviations and identifying a faulty cylindrical system in which the combustion process had ceased. Such an instrument might be mounted directly to the head of a crankshaft, as was done during the present work. Alternatively, the measurement could be carried out on the shaft line or with a device parallel to the crankshaft axis using drive belts or optical measurements by means of a belt pulley and optical encoder.

The rotational speed is one of the basic engine operating parameters and is monitored by tachometers and power measurement systems that are installed as a standard in combustion engines [1,2,22]. Using signals in the method proposed in this paper would enable a more complete assessment of the technical condition and, therefore, contribute to improving the safety and efficiency of the operation process.

Lower-power marine engines are mostly devoid of measuring equipment, which makes diagnostics and engine monitoring very difficult and sometimes impossible. Combustion pressure sensors are also subject to frequent failures due to high temperatures and pressures, so their suitability may be largely limited to double-stroke main drive engines. The accessibility to the drive shaft is another limitation, which is limited in the case of 4-stroke power generators. The solution proposed by the authors was to supervise the combustion process by measuring instantaneous angular velocities using an incremental encoder mounted to the engine crankshaft.

Additionally, in the near future, in consequence of emission limitations, the market would be supported with a huge quantity of dual-fuel gas engines, which are more exposed to knocking combustion in chambers. Gas fuel supplied with air by an inlet air collector and the reaction during combustion are still not fully controlled. Our system could be very helpful in detecting faulty combustion. At present, marine four-stroke dual-fuel gas engines are not equipped with any system for combustion monitoring, and our system could make the engine room safer.

Let  $n$  and  $m$  are variables equal to a number of subsequent crankshaft rotations used for the diagnostic algorithms for steady-state and transient state analysis appropriately. The authors constructed an experimental measurement system and conducted tests to verify the following research hypotheses:

- For a steady-state (quasi-static) engine, the difference in instantaneous rotational speed compared to the average rotational speed of a specific crankshaft position and calculated for  $n$  subsequent revolutions will reflect the change in the steady-state engine condition.
- The deviation from the calculated average for full revolutions  $m$  of the engine's instantaneous rotational speed for each position of the crankshaft allows the technical condition of the engine in the transient state to be evaluated and makes it possible to identify the cylinder system, causing the reduction in engine performance.
- For a given engine and load, there are optimum  $m$  and  $n$  values that allow sufficient assessment for practical changes in the load and technical condition of the engine.

In addition, the experiment aimed at confirming the performance of the designed prototype by identifying the cylinder in which the combustion malfunction occurred. The authors presented the prototype system using the Arduino based I/O system with National Instruments NI-USB-6210 card. The solution presented in the paper was filed for patent protection and now has the status of national and European patent applications [49,50].

The presented experiment showed the applicability of the proposed system, so the authors had the grounds to build the next generation system, which would be completely based on Arduino or an alternative controller. The next-generation prototype was planned to be built with an Armega 2560 microcontroller (2560 16CUR, Microchip Technology, Chandler, AZ, USA) installed on Arduino Mega 2560 (Rev. 3, Arduino, Turin, Italy) or Arduino Industrial 101 board (101 LGA module, Arduino, Turin, Italy).

### 3. Materials and Methods

#### 3.1. Test Subjects and Measuring System

Tests were carried out using a Buckau-Wolf R8DV-136 engine (R8DV-136, VEB SKL, Magdeburg, Germany) directly driving a fixed-pitch ship propeller and a Sulzer 5 BAH 22 engine driving (Sulzer 5 BAH 22, ZUT Zgoda, Świętochłowice, Poland) an alternating current generator loaded with a water resistor. The technical details of both engines are given in Tables 2 and 3.

**Table 2.** The basic technical and operational parameters of a Buckau-Wolf R8DV-136 Engine.

Parameter	Description
Manufacturer	VEB SKL – Magdeburg
Power Consumer	Fixed pitch propeller
Type	4-stroke, naturally aspirated, trunk engine
Number of Cylinders	8
Cylinder Bore	240 mm
Piston Stroke	360 mm
Cylinder Working Volume	16.290 dm <sup>3</sup>
Compression Chamber Volume	1.205 dm <sup>3</sup>
Nominal Effective Power	220 kW
Nominal Speed	360 rpm
Mean Piston Speed	4.32 m/s
Nominal Specific Fuel oil Consumption	238 g/kWh

**Table 3.** The basic technical and operational parameters of a Sulzer 5 BAH 22 Engine.

Parameter	Description
Manufacturer	ZUT Zgoda
Power Consumer	Alternator
Type	4-stroke, turbocharged, trunk engine
Number of Cylinders	5

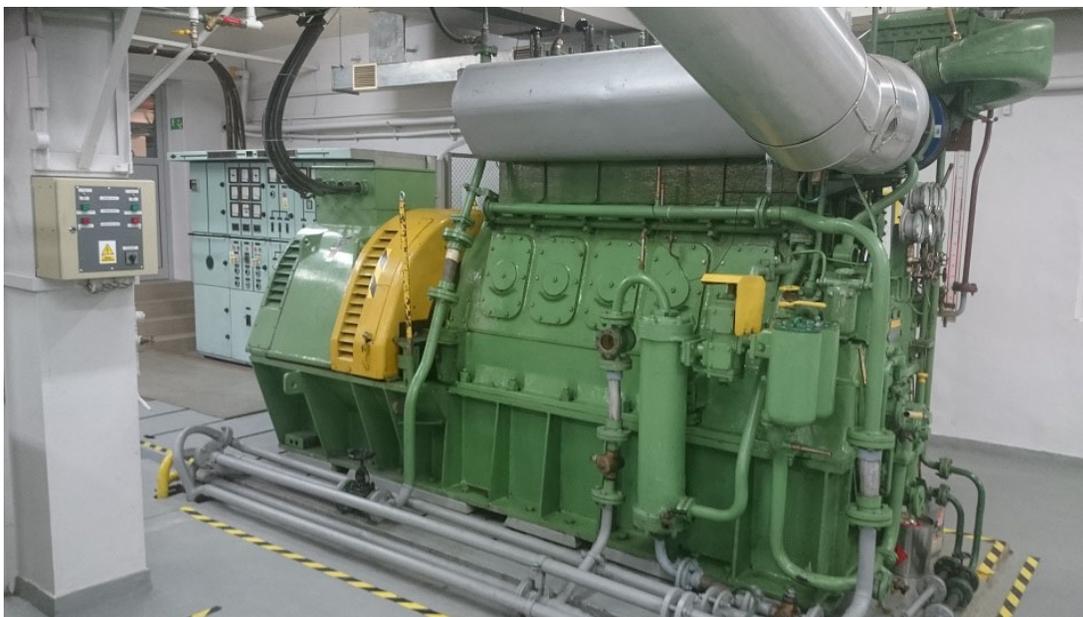
Cylinder Bore	220 mm
Piston Stroke	320 mm
Nominal Effective Power	220 kW
Nominal Speed	500 rpm
Mean Piston Speed	5.33 m/s
Nominal Specific Fuel Oil Consumption	232 g/kWh

The Buckau-Wolf R8DV-136 engine that was pre-tested is shown in Figure 1. Due to the high degree of the speed uniformity reaching 5% and the possibility of quickly suspending fuel pumps using the engine spanner (a dedicated tool for the fuel pump suspension), the engine is a very good research object for assessing changes in instantaneous rotational speed. The encoder for the contour position of the crankshaft was installed at the free end of the shaft.



**Figure 1.** View of the analyzed Buckau-Wolf R8DV-136 engine.

The second engine used during measurements was the Sulzer 5 BAH 22 engine, shown in Figure 2. Power generators have tighter requirements for the permitted instantaneous speed variation compared with main propulsion engines, affecting the correct synchronization conditions and the quality of the produced electricity. These features were the reason for selecting this engine as a test subject to verify the proposed method for diagnosing malfunctions in engine combustion processes.



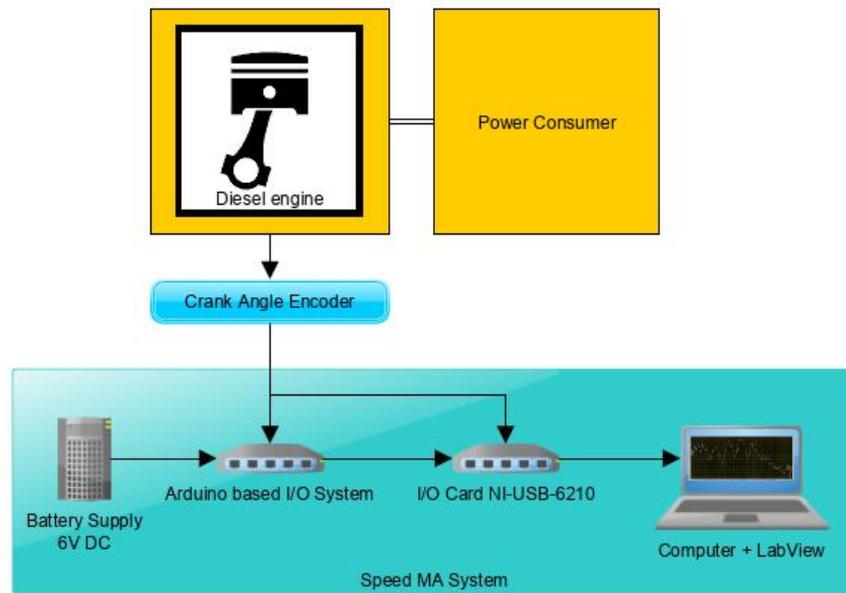
**Figure 2.** View of the analyzed Sulzer 5 BAH 22 engine.

Fuel system malfunctions were simulated by opening the fuel vent valve at the injection valve affected by the malfunction. An engine crankshaft angle encoder was connected to the shaft using a factory threaded mounting hole in the shaft, as shown in Figure 3.



**Figure 3.** View of the BAH engine with a connected measuring device.

Figure 4 shows a diagram of the measurement system, which consists of a National Instruments USB (Universal Serial Bus) measurement card (NI-USB-6210, National Instruments, Austin, TX, USA) and an Arduino Mega 2560 microcontroller (Rev. 3, Arduino, Turin, Italy) integrated with an Arduino UNO board (UNO, Arduino, Turin, Italy).



**Figure 4.** Diagram of the SpeedMA measuring system.

Additionally, the microcontroller had its own 6 V power supply so that the control signal from the encoder did not shift due to the disappearance of the TDC (top dead center) indication of the first cylinder due to turning the computer program off/on. The data collected by the system was processed and visualized using LabView 2018 software (2018 version, National Instruments, Austin, TX, USA)

The encoder had a resolution of 360 samples per revolution, and an additional control signal emitted once per revolution. According to the authors' tests, this was sufficient to properly investigate the phenomena occurring during the engine combustion process up to 2000 rpm, i.e., all 2-stroke and 4-stroke engines used on ships. The four-stroke engine was characterized by two crankshaft rotations per cycle, which required the introduction of an additional impulse that correctly marked the beginning of the cycle on the first of the cylinders. This pulse could be bypassed with software at the time of installation on a two-stroke engine.

An Arduino microcontroller with its own power supply was installed in the measuring system with the USB card. By continuously interfacing with the engine, with the assistance of the algorithm, the microcontroller accurately marked the position of the shaft, through which the algorithm determined the start of combustion in each of the cylinders and the ignition sequence of the engine. The measurement algorithm created in the LabView 2018 environment (2018 version, National Instruments, Austin, TX, USA) was in operation when information about the beginning of a new cycle appeared by closing the corresponding circuit (pin 11). Additionally, in order to facilitate proper interfacing with the engine, two LEDs (light-emitting diodes) were installed in the measuring system to provide information about the presence of a control pulse during each revolution (pin 8) and every two revolutions (pin 10). The signal generated by Arduino lasted 50 ms, then the microcontroller was reset and waited for a new control signal from the encoder. The program controlling the microcontroller was written in C++ and is included in Appendix A.

The graphic code for LabView 2018 was used. Two DAQ (data acquisition) modules from the Lab-View library were used as input signals. Noise or signal distortion was removed by means of a six-points-based moving average filter. One main input that controls the main program loop was

controlled by a 0°CA (TDC) signal in the first cylinder. The impulse signal from the crankshaft pivot points, on the other hand, forced the time reading from the card clock. A computer equipped with an algorithm enabled the continuous supervision of the impulse time, after which the transformation was converted into an instantaneous speed, which directly corresponded to the combustion process occurring in the combustion chambers. It also allowed the waveforms to be saved to a text file, so they could be analyzed later.

### 3.2. Measurement Data Processing Algorithm

To localize malfunctions during the combustion process, a signal processing algorithm in the SpeedMA system was used, which was implemented as a LabView 2018 program, whose description and graphic form are presented below. The algorithm calculated the time between the encoder impulses and determined the value of the real instantaneous speeds between impulses for the entire operating cycle.

$$t_x = i_x - i_{x-1} \left( \frac{\text{ms}}{\text{sample}} \right) \quad (4)$$

$$n_x = \frac{1}{t_x \left( \frac{\text{ms}}{\text{sample}} \right)} \cdot 1000 \left( \frac{\text{ms}}{\text{s}} \right) \cdot 60 \left( \frac{\text{s}}{\text{min}} \right) \cdot 2 \left( \frac{\text{rev}}{720 \text{ samples}} \right) \text{ (rpm)} \quad (5)$$

where:  $t_x$ —time for a given encoder position,  $i_x, i_{x-1}$ —impulses sent by the encoder,  $n_x$ —instantaneous rotation speed.

After an entire cycle, the average instantaneous rotational speed and the deviation from the value for each of the collected measurements were calculated:

$$\bar{n} = \frac{\sum_{i=1}^{720} n_x}{720} \text{ (rpm)} \quad (6)$$

$$\Delta n_{div x} = n_x - \bar{n} \text{ (rpm)} \quad (7)$$

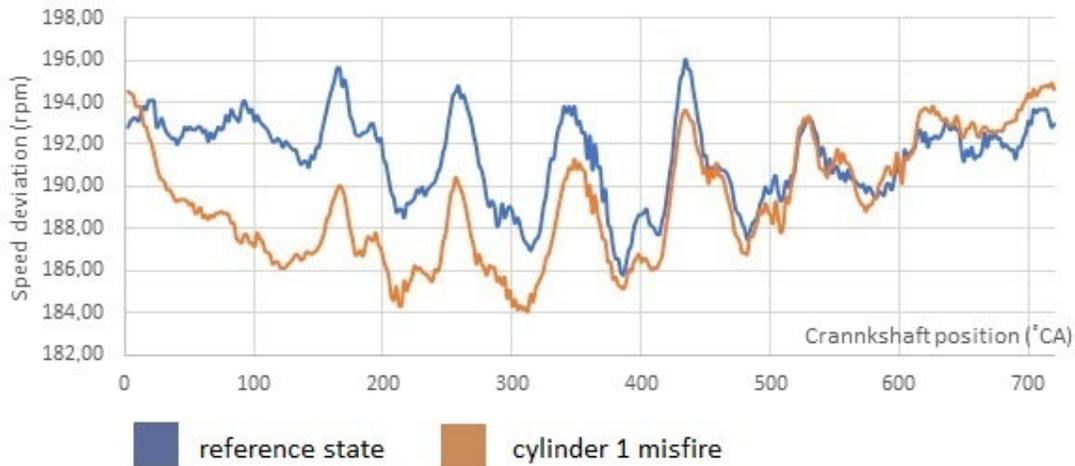
where:  $\bar{n}$ —average instantaneous rotation speed for the cycle,  $\Delta n_{div x}$ —the deviation from the average value for each of the measurements over the cycle.

To correctly evaluate the deviation of individual cylinders, the value of the crankshaft rotation angle should be determined, the range of which includes the maximum deviation, which would enable the monitoring software to compare it with the limit value to indicate the defective cylinder. For the purpose of the algorithm, this value was called the  $W$  coefficient. This value would vary depending on the type of engine, the number of strokes or cylinders, speed controller characteristics, and the condition of the engine itself and its design. Based on our own measurements, we assumed that the angle in the cylinder number function would take the following general form, which should be estimated during calibration using a real engine test/survey:

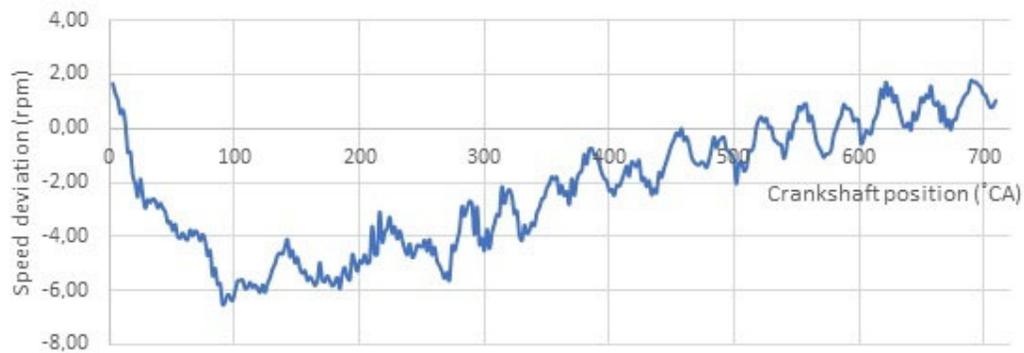
$$W = \left( \frac{1}{2} + \frac{1}{i} \right) C \text{ (°CA)} \quad (8)$$

where:  $i$ —cylinder count,  $C$ —the number of measurements in a full operating cycle, depending on the number of piston strokes per 1 operating cycle ( $C = 360$  for a 2-stroke engine, and  $C = 720$  for a 4-stroke engine).

In order to determine the exact  $W$  coefficient for a given engine, a calibration measurement must be carried out (Figure 5), which should take place under the conditions set for each of the possible operating loads. Then, a controlled malfunction should be introduced in the form of a loss of injection process in a known cylinder. Depending on the engine type, this could be accomplished by suspending the fuel pump, opening the bypass valve, and setting the fuel dose to "0" on the selected cylinder. The collected data, in the °CA function, should be combined with each other by calculating the instantaneous deviation caused by the malfunction (Figure 6). For complete data collection and averaging, a similar measurement should be carried out for each engine cylinder.



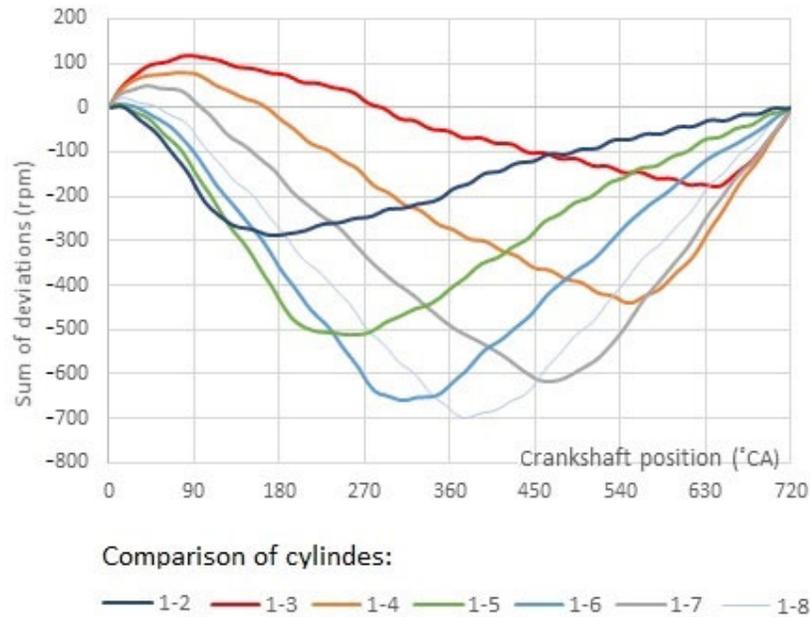
**Figure 5.** Reference instantaneous Buckau-Wolf R8DV-136 engine speed for the pre-set speed of 190 rpm.



**Figure 6.** The waveform of instantaneous angular velocity deviation of the engine Buckau-Wolf R8DV-136 for a pre-set speed of 190 rpm with no combustion in cylinder 1.

The value sought by the algorithm was the maximum sum of negative values of the defective cylinder. The location where the highest drop in speed for a given system  $x$  occurred indicated a defective cylinder or one which was not fully operational. In the described calibration process, the simulated cylinder No. 1 was defective, so further analysis and selection of the appropriate calculation range would be carried out based on it. The authors proposed one of two methods for selecting the  $W$  coefficient: a graphical method illustrating simulations for the entire cycle range or a mathematical method.

When choosing the graphical method, the sum of the deviations of cylinder 1 should be compared with the sums calculated for each of the other cylinders as a crankshaft rotation function. The sought-after value was a crank angle, which represented the maximum negative deviation between cylinders and indicated the highest speed drop among all cylinders. The course of the simulation is shown in Figure 7. According to the chart below, each  $W$  value 288–719°CA was within the range, but the largest total deviation for all cylinders was around 445°CA. This value was close to the value of 450°CA, calculated according to formula No. 3, which made it possible to apply a  $W$  coefficient of 450°CA.

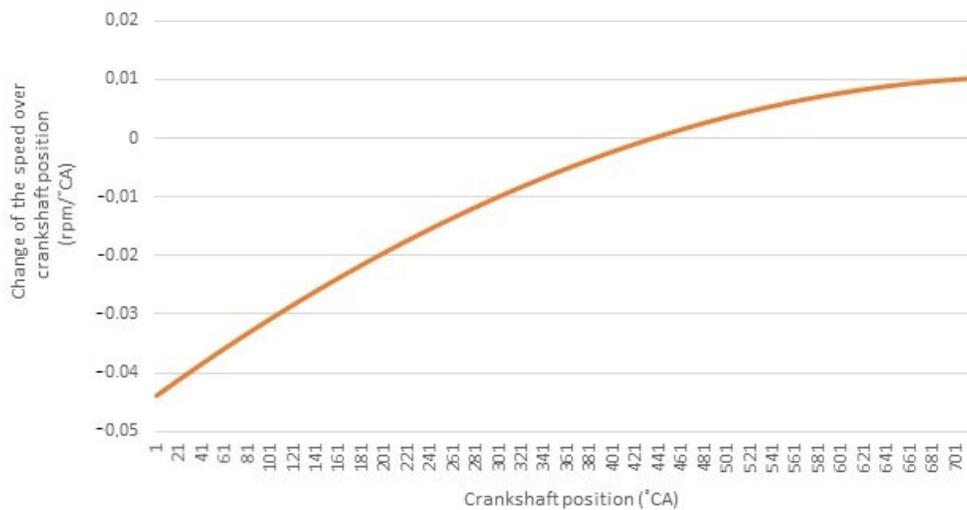


**Figure 7.** Summary deviations of rotational speed with a crankshaft rotation function for a Buckau-Wolf R8DV-136 engine.

The mathematical calculation used to determine the interpolating function for a graph illustrating simulations and for solving the equation digitally or analytically was an alternative to a graphical representation:

$$\frac{dn}{da} = 0 \text{ (rpm/rad)} \tag{9}$$

By consecutively determining the maximum deviation values, we determined the angle at which the deviation reached its maximum (Figure 8). The result of the calculation was a graph and the resulting value of 435°CA, which was close to the value from Equation (3). Therefore, the W value of 450°CA would be used for further calculations.



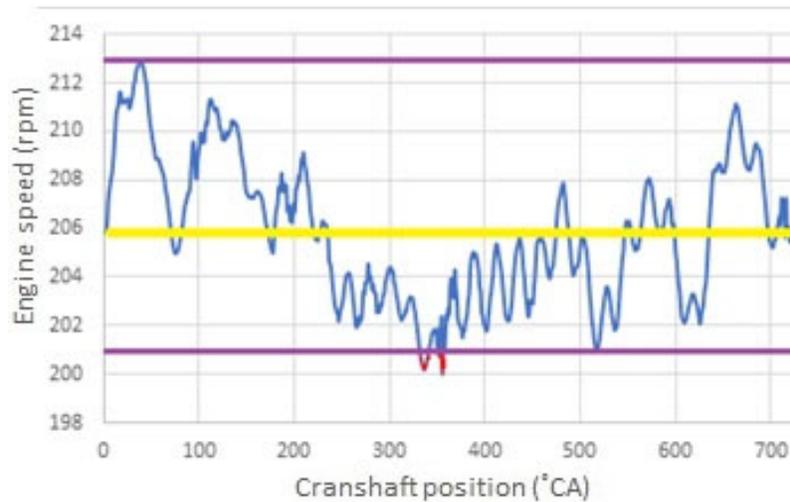
**Figure 8.** Determination of the maximum deviation angle of a Buckau-Wolf R8DV-136 engine obtained by a differential equation.

Knowing the value of the  $W$  coefficient allowed the algorithm to calculate the sum of speed deviations within its range after each cycle, according to Equation (4). The resulting values were comparable to the reference values, which were continuously updated after each valid cycle according to Equation (5). If the deviation of the real value exceeded the set value  $n_x$  in the program, the event was indicated as an alarm for the cylinder  $x$  equal to.

$$n_{x\ cyl} = \frac{\sum_{i=1}^5 (\sum_{\varphi \in [TDC, TDC+W]} n(\varphi))}{5} \quad (\text{rpm}) \quad (10)$$

where  $i=1-5$  subsequent engine cycles,  $n(\varphi)$  speed at crank angles in the range between TDC in a given cylinder and angle equal to TDC+W.

Simultaneously, the deviation of the real engine was checked using the measured deviation between the real and reference values. If the deviation in the instantaneous rotational speed  $n_{sr}$  from the average  $\bar{n}$  for any of the cylinders was exceeded, an operation irregularity alarm would be activated to inform the user. The example in the diagram in Figure 9 showed an average speed of 206 rpm for the instantaneous rotational speed of the engine. The permissible deviation in the software was set to  $\pm 5$  rpm, and this value was exceeded for the range 330–360°CA. This range corresponded to the combustion process occurring in cylinder 6; therefore, an activated alarm indicated that cylinder 6 was causing an irregularity. The presented algorithm is shown in block form in Figure 10.



**Figure 9.** Graph showing a situation that exceeds the allowed operation deviation.

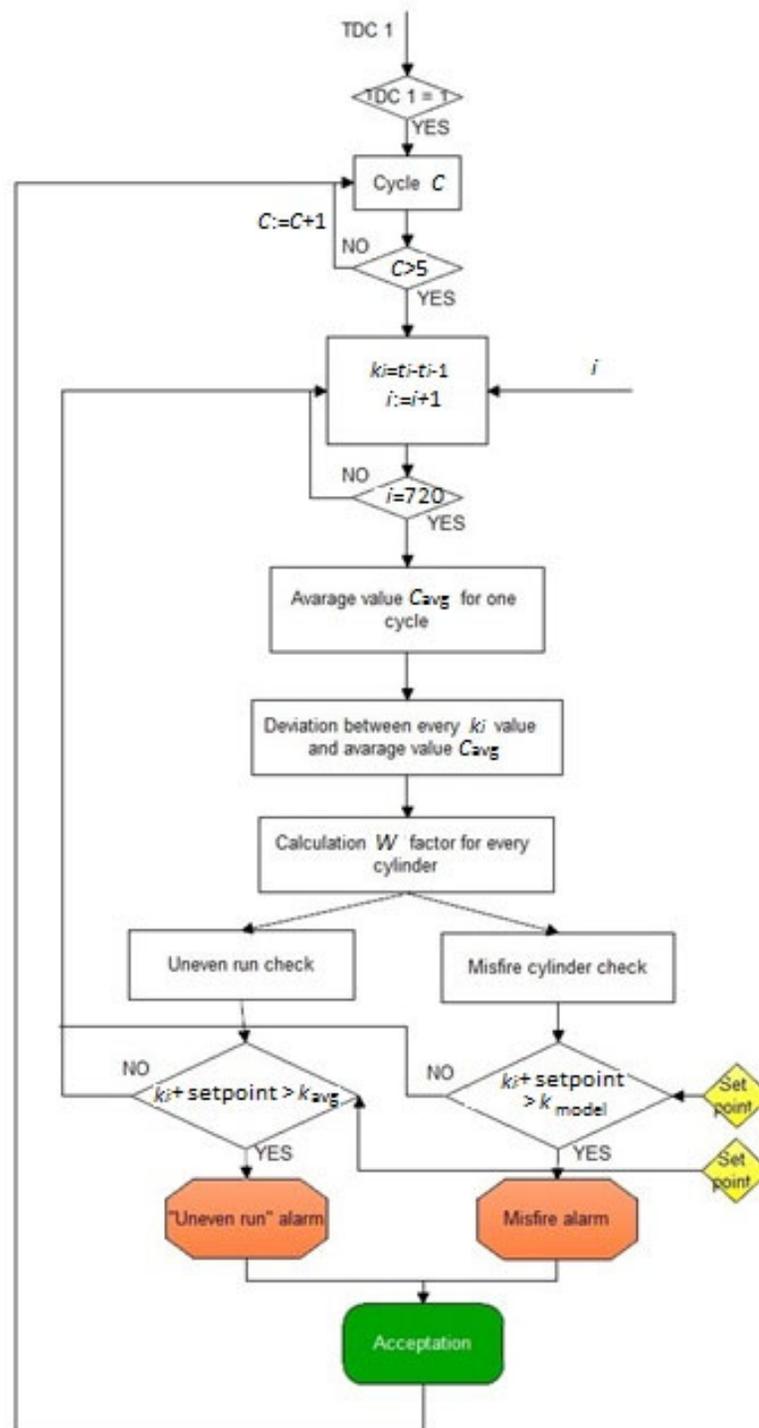
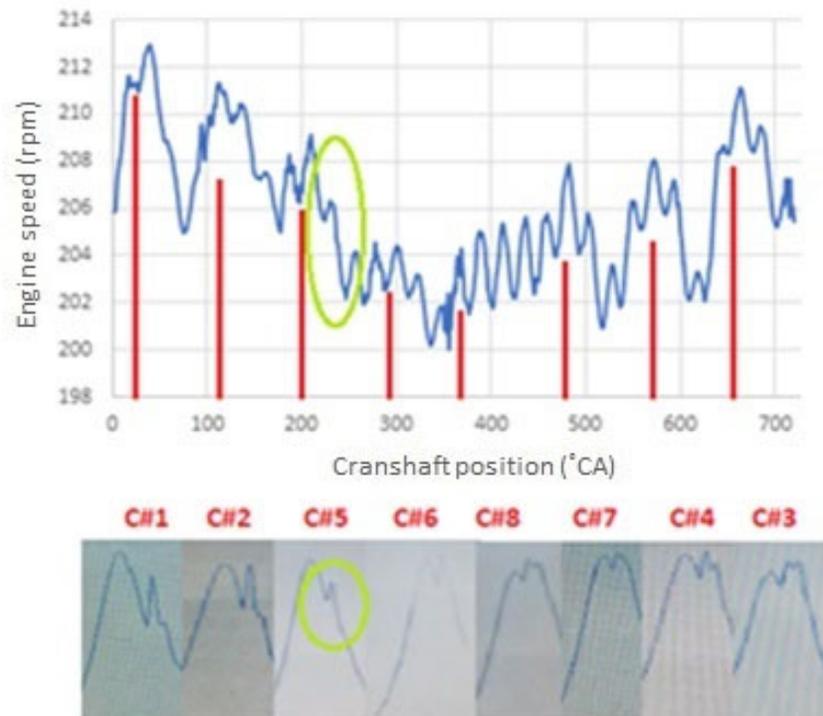


Figure 10. Engine cylinder misfire detection algorithm.

The algorithm operated in a loop, which could be terminated by pressing the "stop" button. Monitoring was activated when the first 6 operating cycles were completed. This was due to the need to create reference values for the first 5 cycles, to which the real values would then be compared. The limit parameters could be changed by changing the setting in the program or by adjustments using potentiometers.

#### 4. Results and Discussion

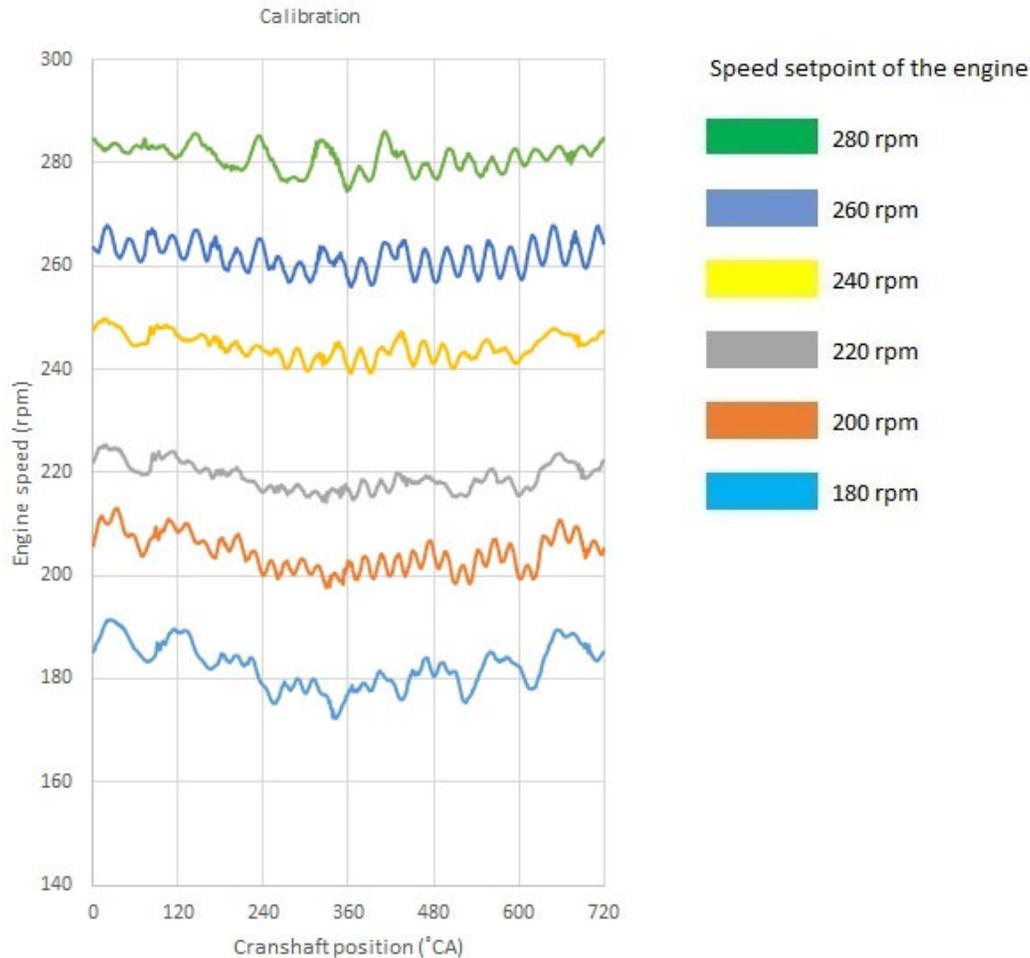
During the initial stage of work on the invention, the authors focused on verifying the relationship between the instantaneous angular velocity and the course of the combustion process. A laboratory motor was simultaneously measured with a combustion pressure indicator and an incremental encoder. Measurements confirmed the occurrence of characteristic elevations at the moment of combustion of the fuel-air mixture, the frequency and proportions of which coincided with each other in the two diagrams presented in Figure 11.



**Figure 11.** Example of the instantaneous rotational speed and combustion pressure as a function of crank angle.

Instantaneous speed dropped near cylinders No. 5 and 6, which corresponded to the indicator charts, and since the increasing of the combustion pressure in these cylinders appeared to be the lowest. The engine being tested had a shifted (delayed) fuel injection due to the reverse operation, causing the combustion curves to deviate from those typical for a diesel engine. Nevertheless, even in this case, it was possible to observe an uneven load on the engine, which coincided with changes in the instantaneous rotational speed. For the sake of transparency of calculations and easy interpretation of the results, the instantaneous angular velocity was changed to instantaneous rotational speed throughout the remainder of the article.

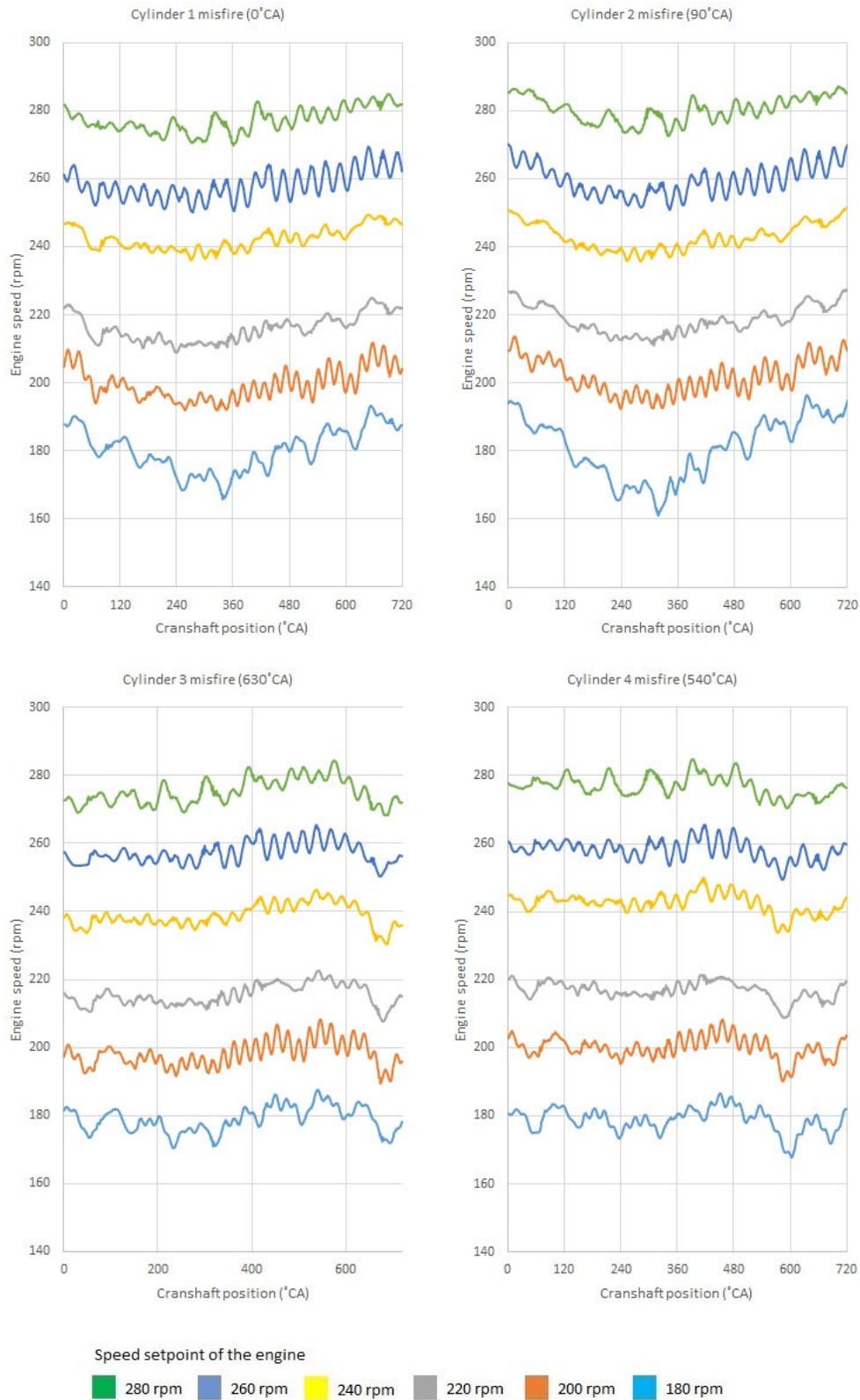
The relevant part of the experiment began by collecting the reference values presented in Figure 12.



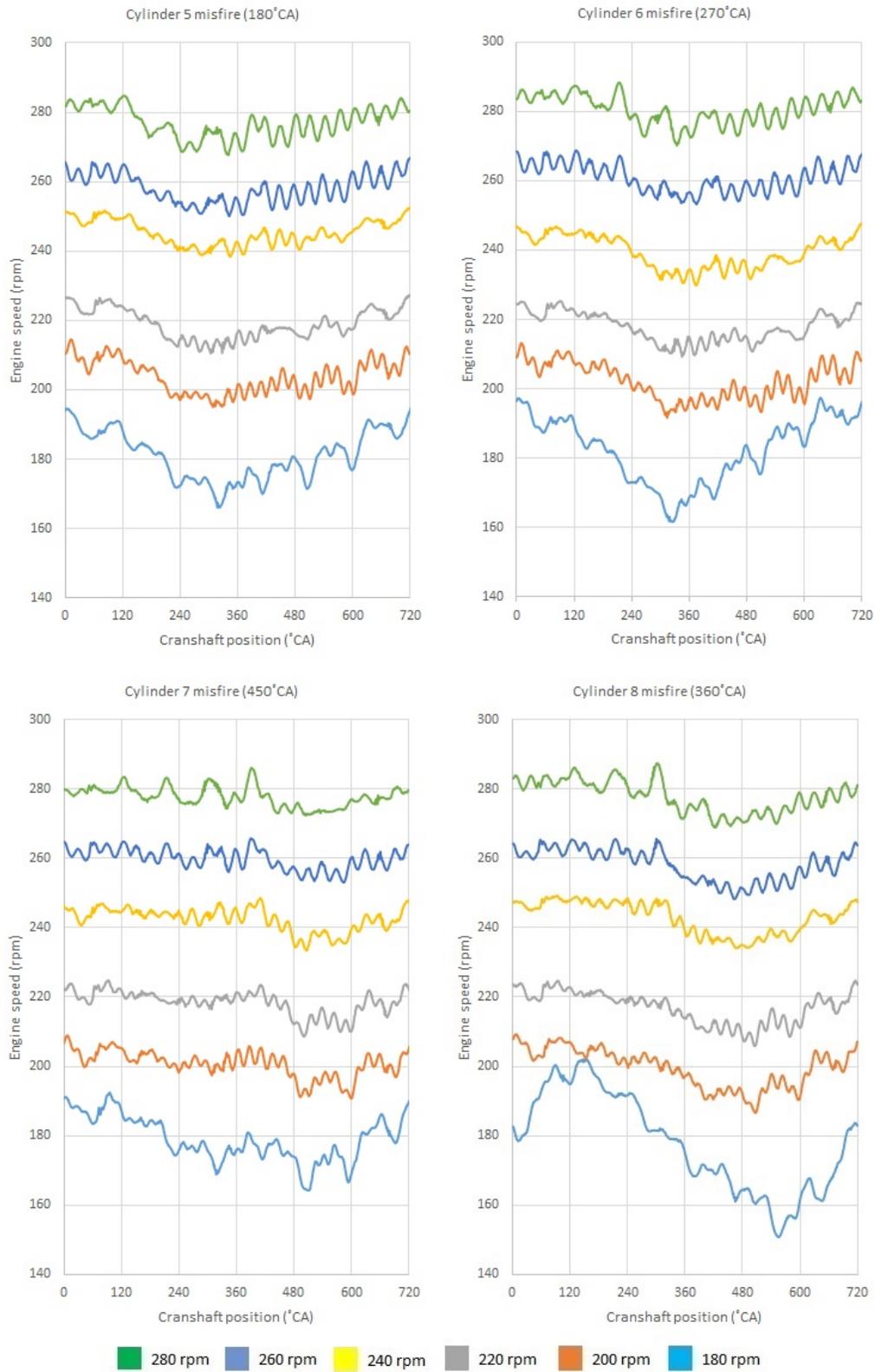
**Figure 12.** Reference measurements for a Buckau-Wolf R8DV-136 engine speeds from 180–280 rpm.

Next, artificial malfunctions of the injection process were introduced in individual cylinders by suspending high-pressure fuel pumps for different engine speed settings, as shown in Figures 13 and 14. The order of Buckau-Wolf R8DV-136 engine ignition during the measurements was constant and involved cylinders 1-2-5-6-8-7-4-3. Ignition in the first cylinder occurred at 0°CA, and each subsequent cylinder was a multiple of 90°CA. The graphs in Figures 13 and 14 showed a significant deviation of the instantaneous rotational speed at the time of the malfunction. This deviation occurred at the point of crankshaft rotation after a malfunction and equalized with the reference value after about 450°CA.

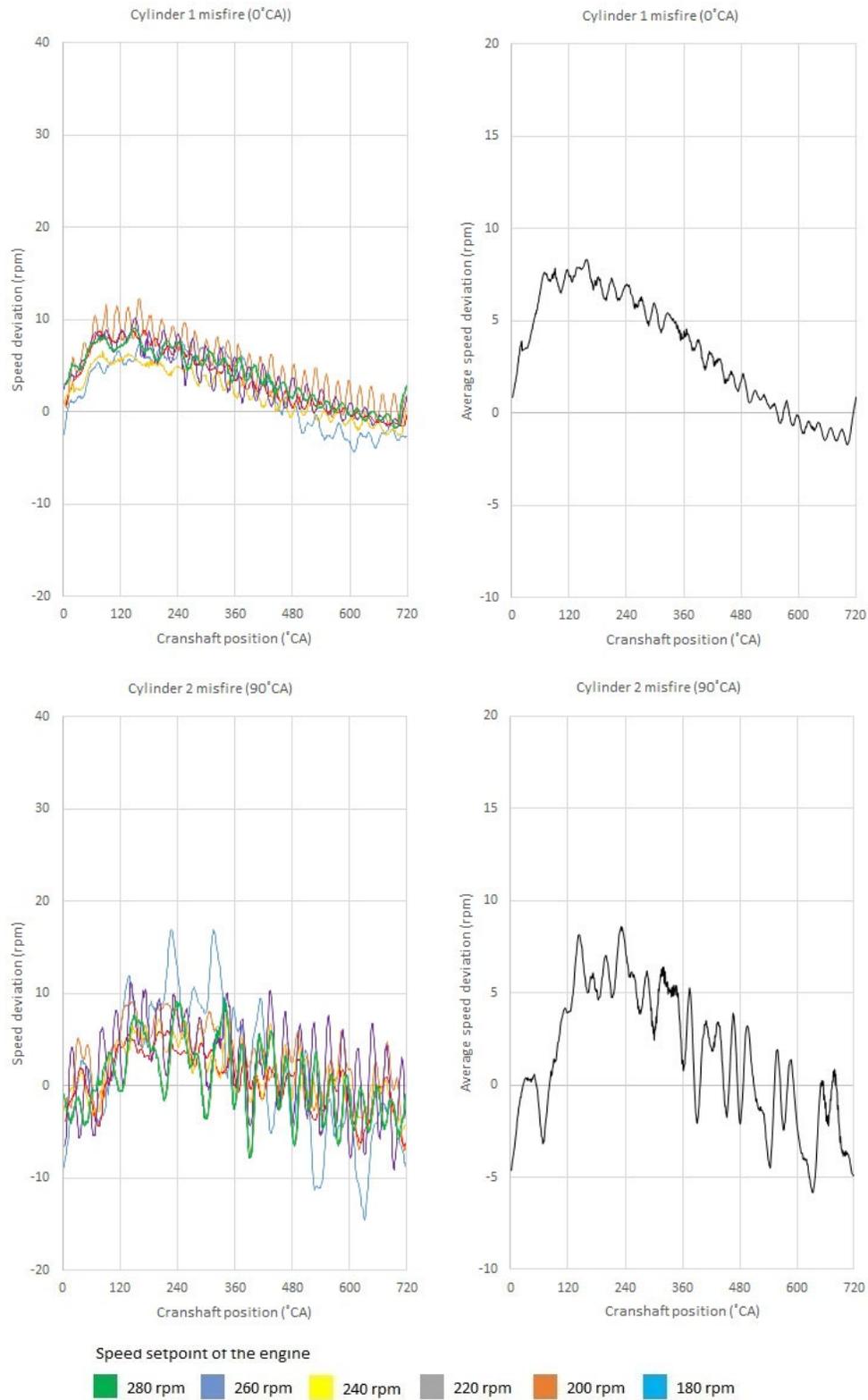
Pursuant to the proposed algorithm, a comparison of the reference values with the values from the waveforms of damaged cylinders was carried out successively in the form of double diagrams for each of the malfunctions. For each cylinder, the waveforms showing the difference (deviation) of the instantaneous rotational speed during malfunction with the instantaneous reference speed and the average deviation for all analyzed speeds were collected. The results are presented in Figures 13–18.



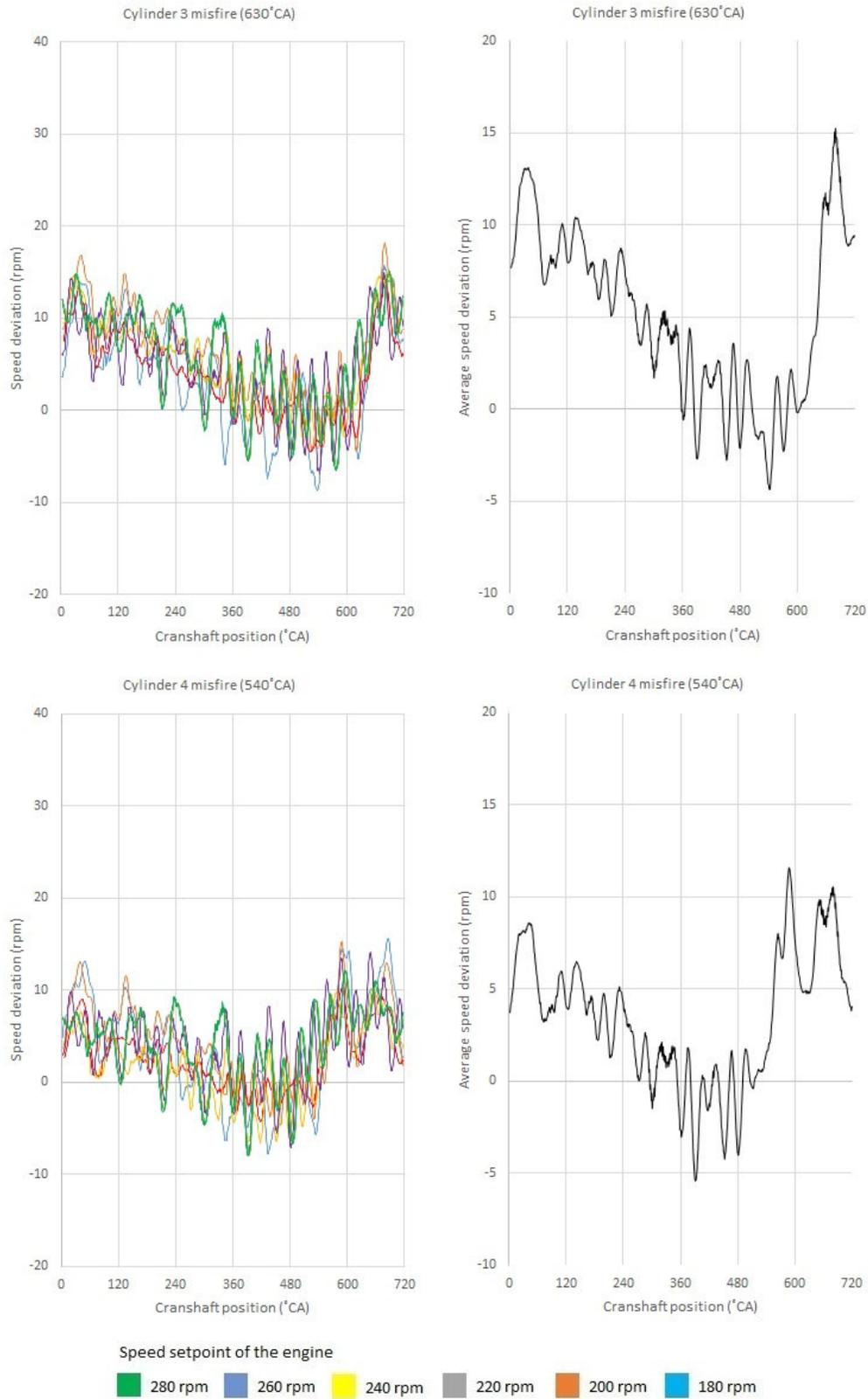
**Figure 13.** Instantaneous speed ranges at the time of combustion loss in cylinders 1–4 of a Buckau-Wolf R8DV-136 engine with pre-set speeds of 180–280 rpm.



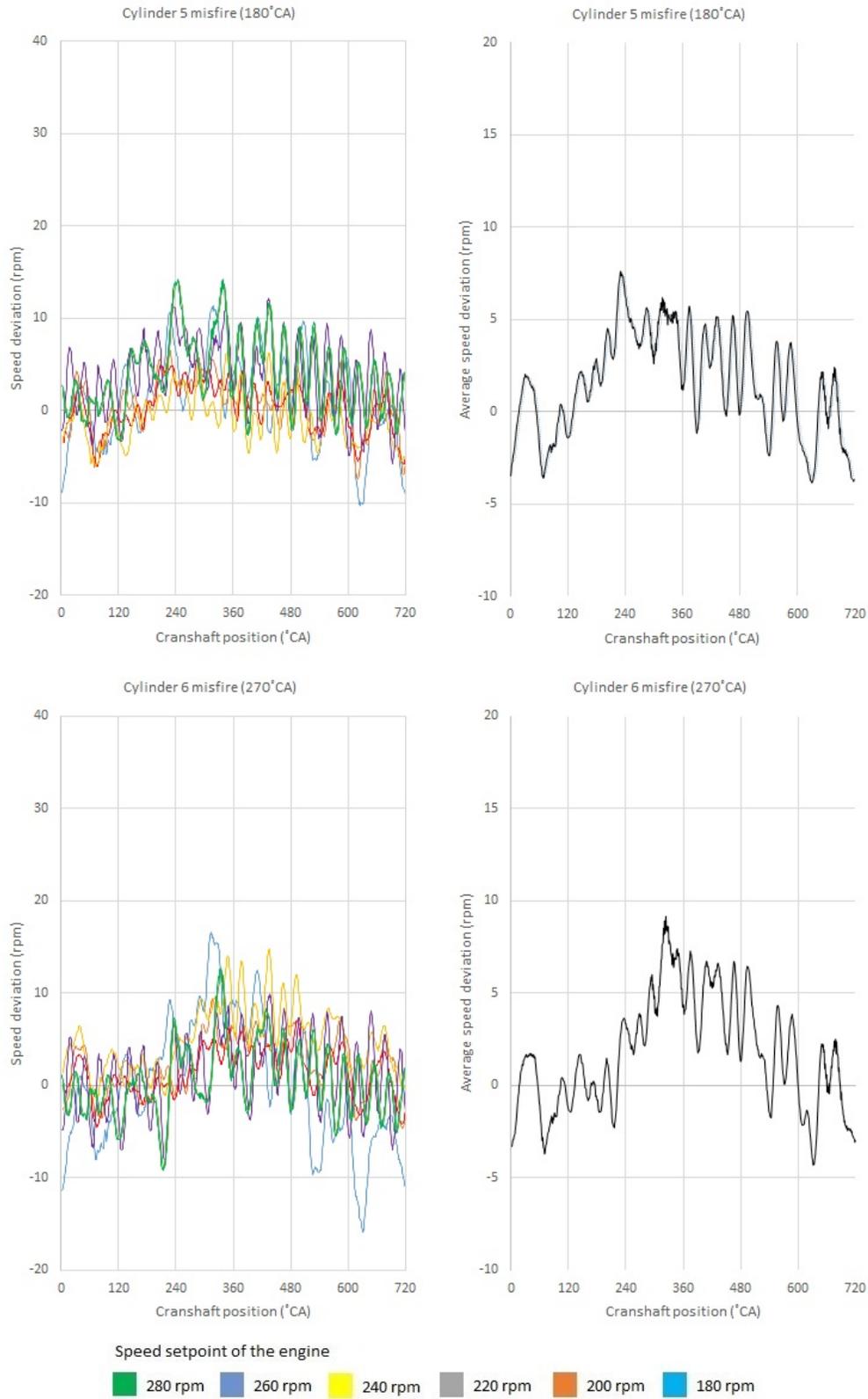
**Figure 14.** Instantaneous speed ranges at the time of combustion loss in cylinders 5–8 of a Buckau-Wolf R8DV-136 engine for the pre-set speeds of 180–280 rpm.



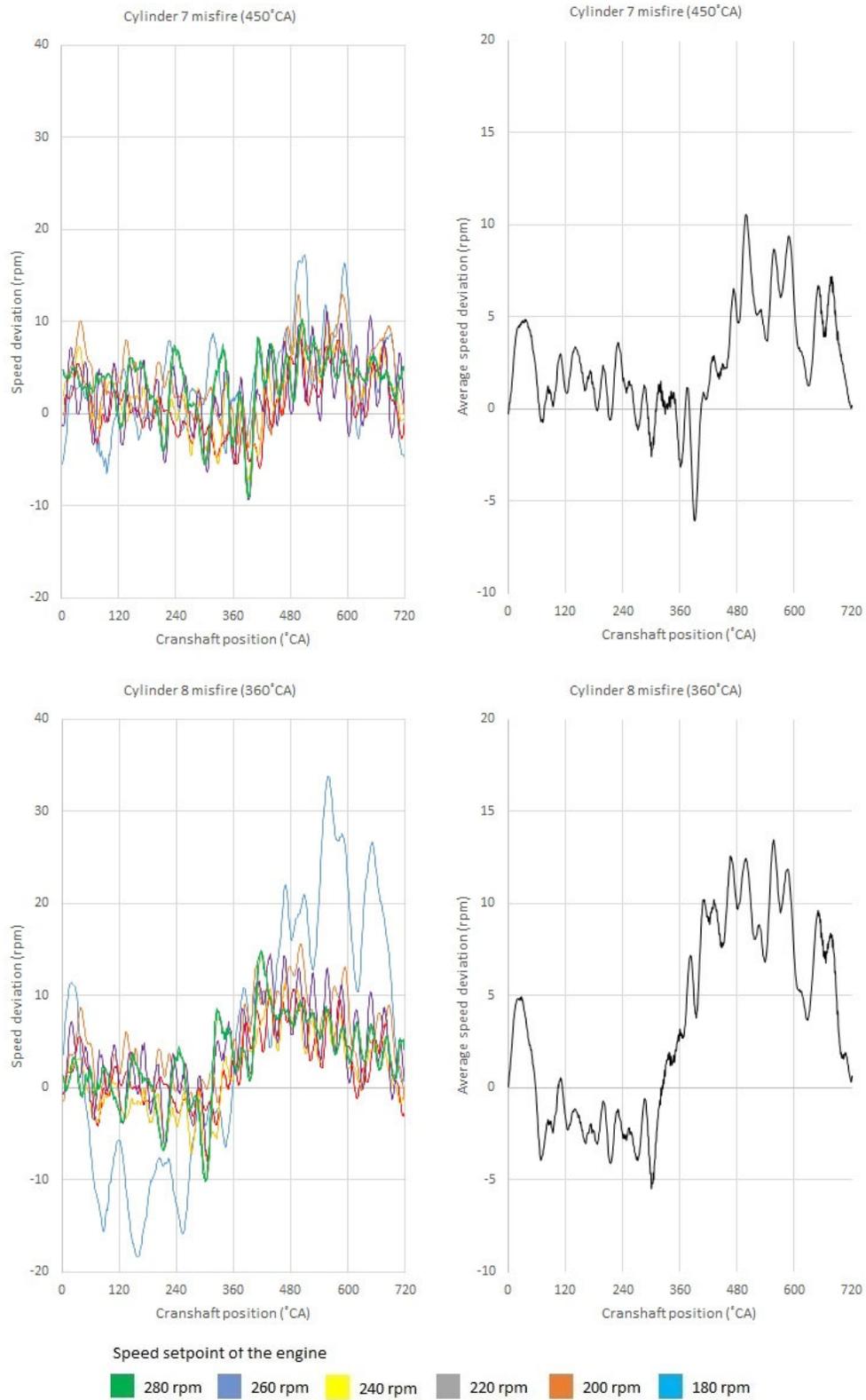
**Figure 15.** Instantaneous speed differences and the average deviation of the combustion loss in cylinders 1 and 2 of a Buckau-Wolf R8DV-136 engine.



**Figure 16.** Instantaneous speed differences and the average deviation of the combustion loss in cylinders 3 and 4 of the Buckau-Wolf R8DV-136 engine.



**Figure 17.** Instantaneous speed differences and the average deviation of the combustion loss in cylinders 5 and 6 of the Buckau-Wolf R8DV-136 engine.



**Figure 18.** Instantaneous speed differences and the average deviation of the combustion loss in cylinders 7 and 8 of the Buckau-Wolf R8DV-136 engine.

The program with a  $W$  coefficient set to  $450^{\circ}\text{CA}$  selected the unsuitable suspending cylinders, correctly informing the user with a signal lamp and information about the misfired cylinder via a user interface. Once the fuel pump of the fourth cylinder was suspended, the deviation in  $W$  exceeded the set limit value, resulting in an alarm.

Measurement was carried out on a power generator, which was analogous to the experiment carried out on the main drive engine. To simulate maritime conditions, a significant droop speed was set, which was used to simulate the readiness of the unit for parallel operation similar to that on marine vessels. A decrease in the rotational speed with an increase in the load is a result of the rotational speed controller, detecting the speed droop function when a single generating unit works with the power network. According to the guidelines of classification associations, the percentage decrease in the speed characteristics should be less than 5% of the whole load range, which is 3%–4% in practice.

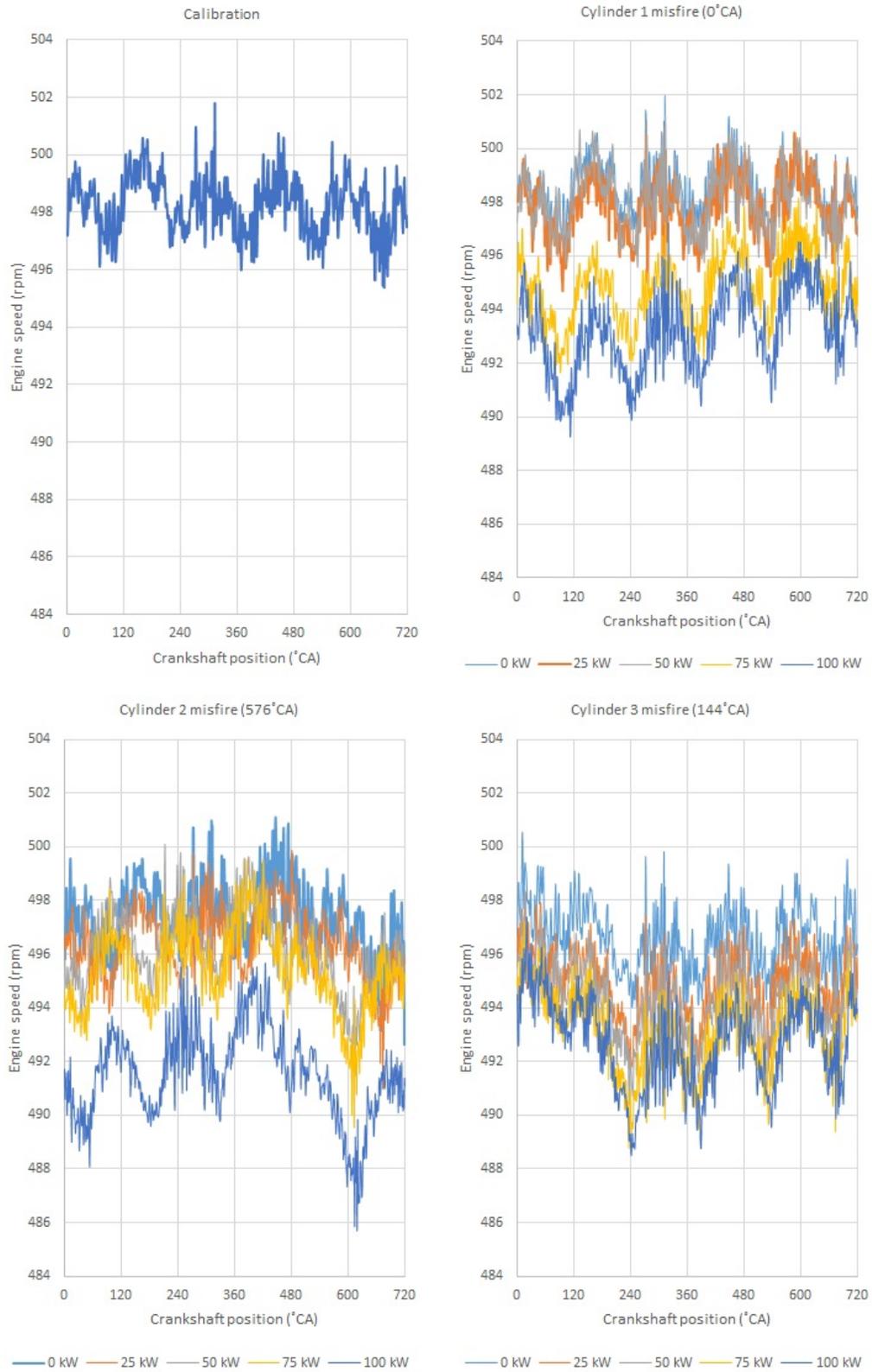
The coefficient  $W$  was determined by the Equation (5) at  $432^{\circ}\text{CA}$ . In the beginning, the standard measurement for the 0 kW setting was recorded, the course of which was represented by black in the diagrams below. Using the diagrams made it easy to read the  $^{\circ}\text{CA}$  values in which the pistons of individual cylinders were located in the TDC. In this instance, these would be points 0, 144, 288, 432, and  $576^{\circ}\text{CA}$ . The instantaneous decrease in rotational speed was even more evident when calculating the difference between the reference and collected speed of a defective cylinder. According to the following graphs, this decrease was significant and could be detected by the working monitoring system. During this test, apart from saving the waveforms to text files, the device constantly supervised the combustion by correctly selecting unfit cylinders.

Similar to the engine operating at a variable speed, the suitability of the proposed algorithm was also tested for the generator set driven by Sulzer 5 BAH 22 engine. The order of engine ignition during the measurements was constant and involved cylinders 1-3-5-4-2.

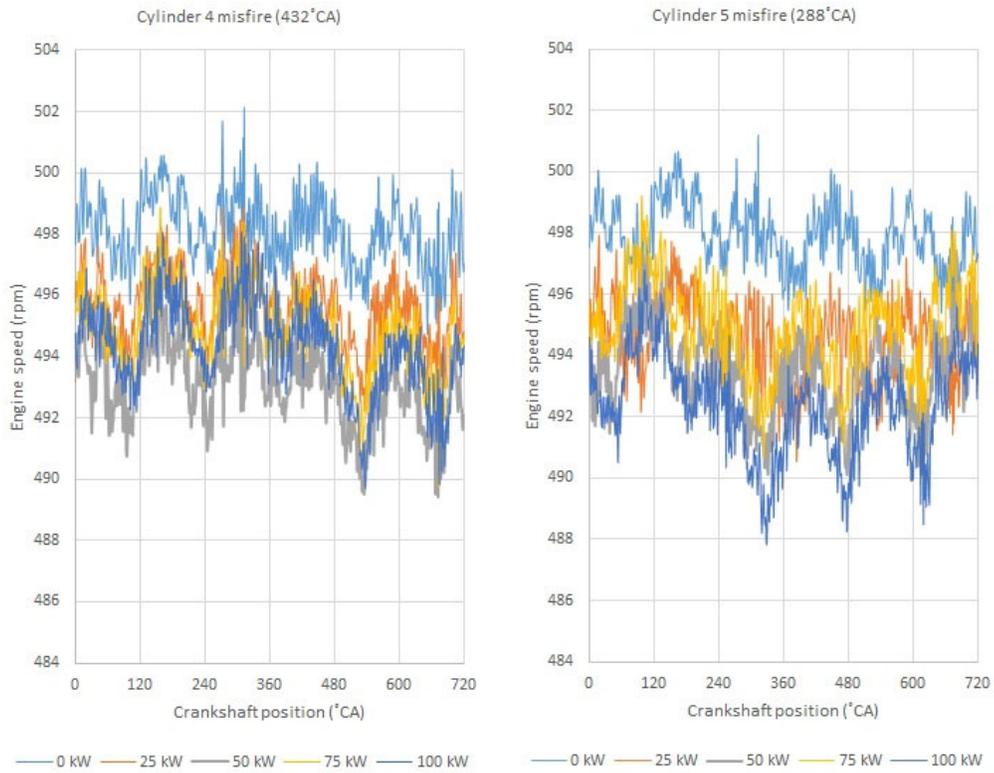
Figures 19 and 20 show the instantaneous engine speed waveforms for a reference state (calibration) and the waveforms with simulated combustion loss in their respective engine cylinders. A single reference run was made for the generator set engine due to small changes in the average speed within the investigated load range. The engine was loaded in the electric power range 0–100 kW with an increased step of 25 kW.

The engine was not loaded above 100 kW after simulated combustion in one of the cylinders in order to prevent damage to the engine due to thermal and mechanical stress on the other cylinders without combustion loss.

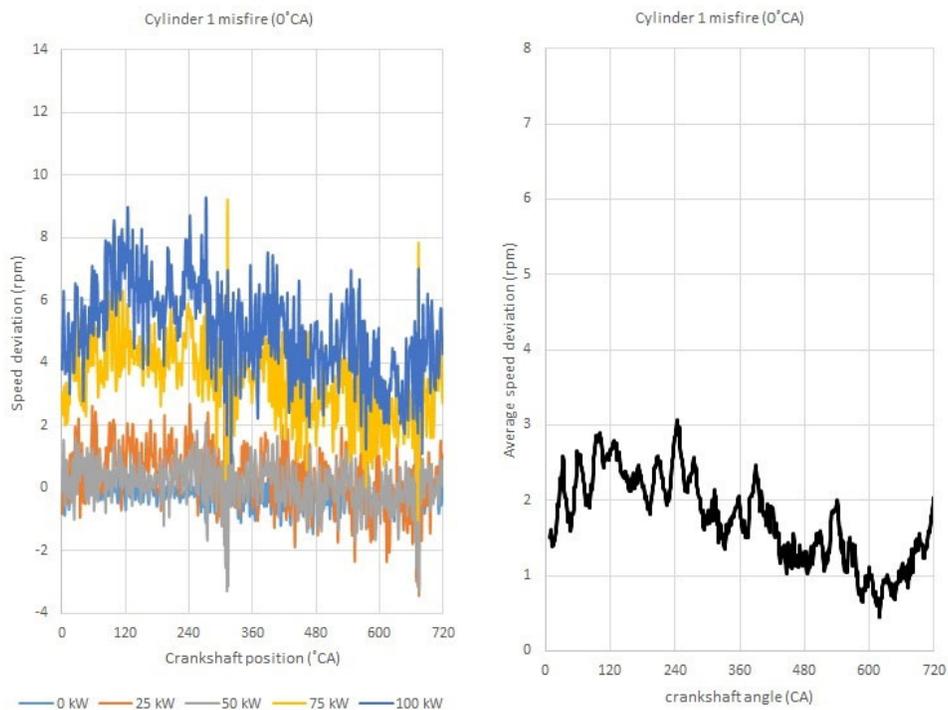
Waveforms showed the difference (deviation) in the instantaneous rotational speed at a malfunction from the instantaneous reference speed and the average deviation for all analyzed speeds. The results are shown in Figures 21–23.



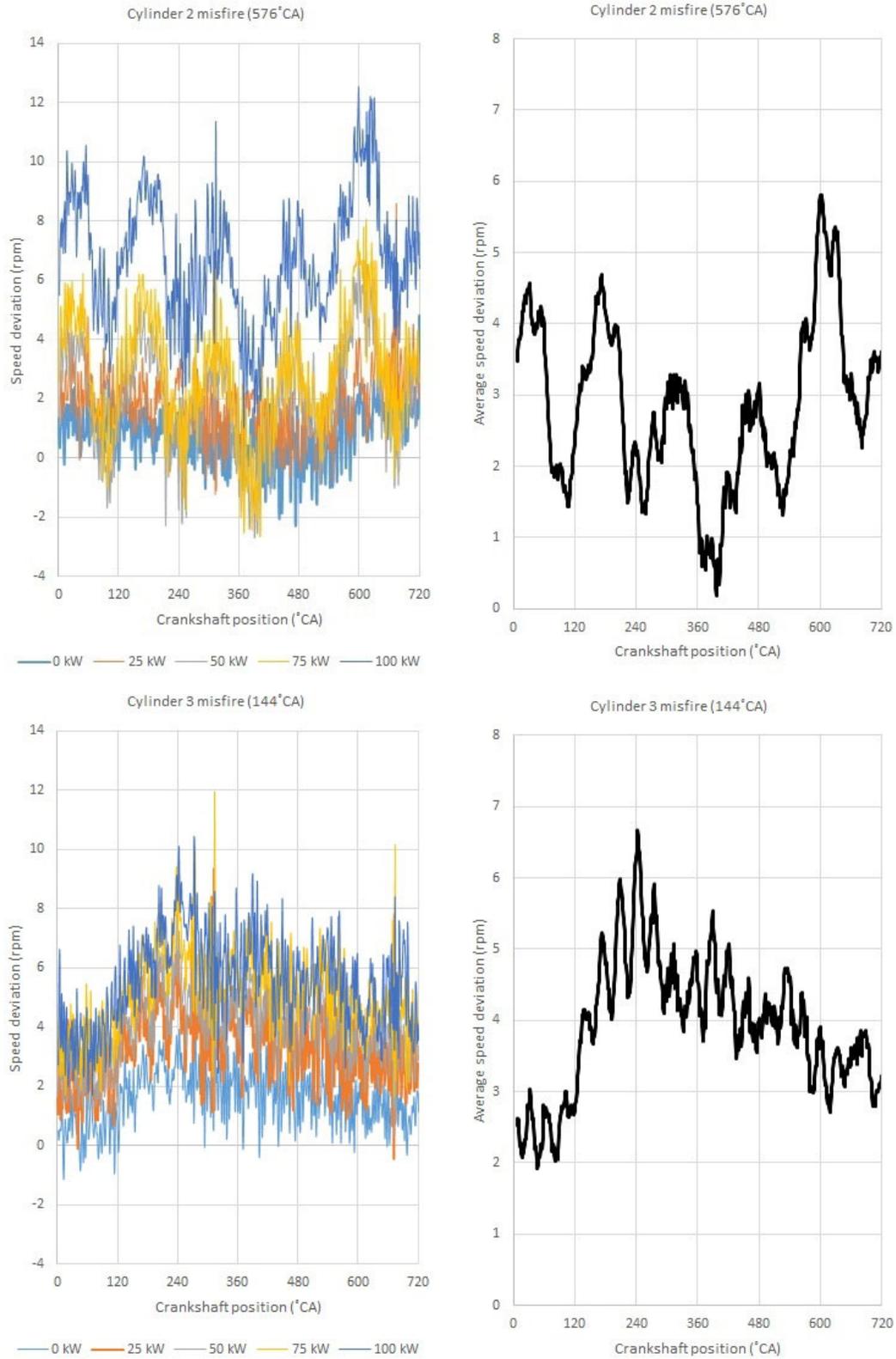
**Figure 19.** Reference measurement and instantaneous speed cycles at the moment of combustion loss in cylinders 1–3 of Sulzer 5 BAH 22 engine for pre-set electric loads of 0–100 kW.



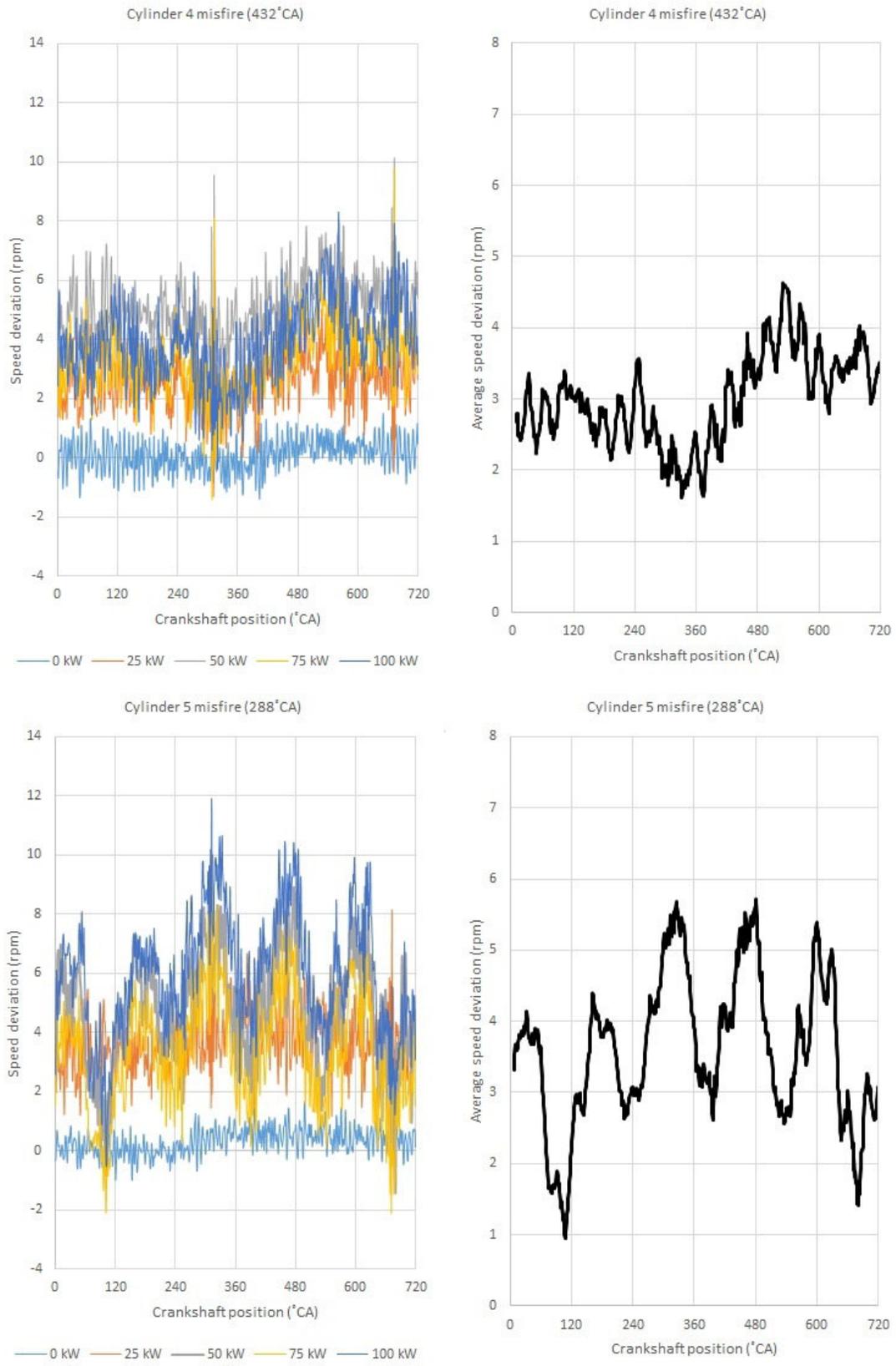
**Figure 20.** Instantaneous speeds at the moment of combustion loss in cylinders 4 and 5 of a Sulzer 5 BAH 22 engine for pre-set electrical loads of 0–100 kW.



**Figure 21.** Instantaneous speed differences and the average deviation of the combustion process in cylinder 1 of a Sulzer 5 BAH 22 engine.

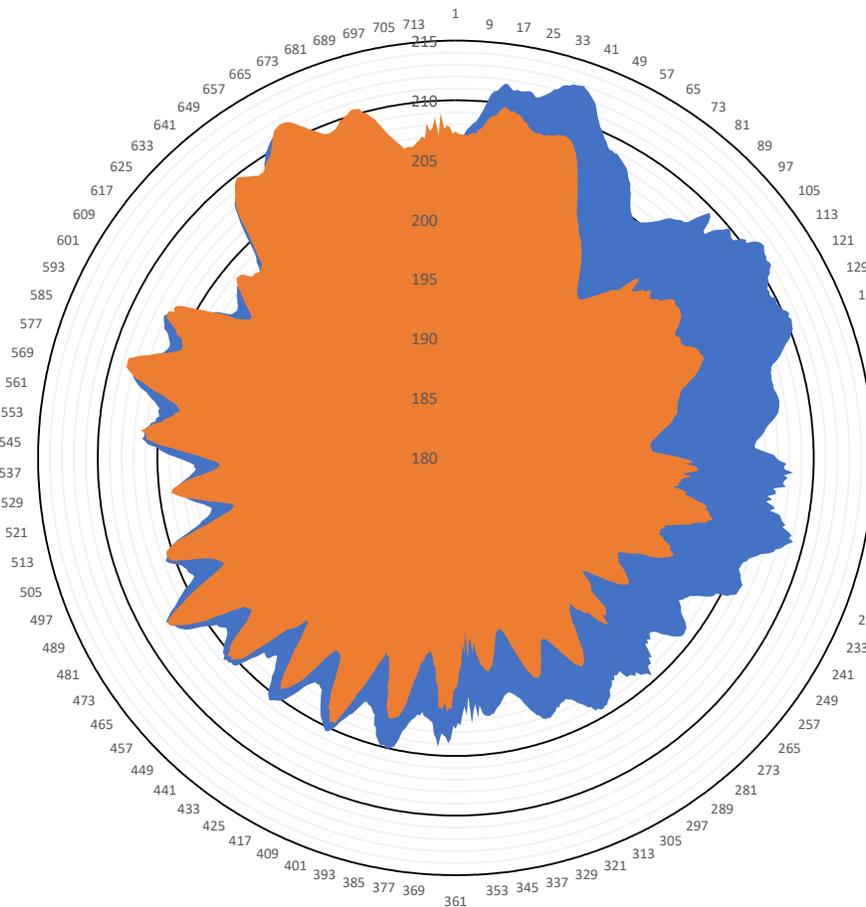


**Figure 22.** Instantaneous speed differences and the average deviation of the combustion process in cylinders 2 and 3 of a Sulzer 5 BAH 22 engine.



**Figure 23.** Instantaneous speed differences and the average deviation of the combustion process in cylinders 4 and 5 of a Sulzer 5 BAH 22 engine.

In order to improve the readability of the analysis results, the waveforms were presented as polar coordinate system graphs. An example of this type of visualization for the Buckau-Wolf R8DV-136 engine is shown in Figure 24, illustrating the situation when the combustion process in cylinder 1 of the engine was lost. The reference values were shown in the form of a blue outline, and the values of the damaged cylinder were shown as the space marked in orange. This was a function of the double rotation of the crankshaft equal to one four-stroke engine operating cycle. The blue field indicated the area that is the sum of the deviations caused by damage to the cylinder; in this case, it was cylinder 1.



**Figure 24.** Changes in the instantaneous rotational speed of the engine shown in the form of a polar coordinate system. Blue: engine in up-state; Orange: cylinder 1 misfire.

The difference in the instantaneous rotational speed from the average rotational speed determined for a specific crankshaft position and calculated for  $n$  subsequent revolutions of the two different examined engines corresponded to a change in the steady-state engine condition. Additionally, the deviation from the instantaneous rotational speed of the engine for each position of the crankshaft from the average calculated for  $m$  full revolutions made it possible to assess the technical condition of the engine.

## 5. Conclusions

A method to determine the coefficient  $W$ , which expressed the differences in instantaneous speed values between standard and defective operations, was a useful method that could be applied in the final version of a device. It was found to be accurate for inline engines with different numbers of cylinders (5 and 8 cylinders). The engine speed controller type and the type of load were shown

not to significantly impact the accuracy of the calculation of the coefficient. Thus, its versatility was confirmed.

The presented results supported the hypotheses made by the authors. Moreover, they showed the practical applicability of a diagnostic system based on the proposed instantaneous rotational speed processing algorithm of the engine to detect combustion loss in a cylinder. Performing an analysis like one on a Buckau-Wolf (R8DV-136, VEB SKL, Magdeburg, Germany) or Sulzer BAH engine (Sulzer 5 BAH 22, ZUT Zgoda, Świętochłowice, Poland) might provide an alternative to other currently available combustion process diagnostic methods. The algorithm, as well as the whole device, correctly indicated the lack of combustion process in an artificially-forced no firing cylinder.

Further efforts should be directed towards allowing the program to identify the type/source of malfunction present in an engine cylinder.

## 6. Patents

1. Chybowski L., Kazienko D., *Method of mechanical load estimation of power machines, preferably piston combustion engines*. European Application EP 19460053.2.
2. Chybowski L., Kazienko D., *The method of mechanical load estimation of power machines, preferably piston combustion engines*. Application to Polish Patent Office P.427649.
3. Chybowski L., Kazienko D., *Electronic device for detecting deviations of the signal from sensors monitoring the state of the diagnosed object and a way of this detection*. European Application EP 19460052.4.
4. Chybowski L., Kazienko D., *Electronic device for detecting deviations of the signal from sensors monitoring the state of the diagnosed object and a way of this detection*. Application to Polish Patent Office P.427648.
5. Chybowski L., Kazienko D., *Universal resistance system supporting the calibration of resistance sensors in energetic machines monitoring systems, preferably internal combustion engines*. Application to Polish Patent Office P.429574.
6. Chybowski L., Kazienko D., *Housing of the detector of diagnostic signals values deviations*. Polish Patent Office registered design 25673.

**Author Contributions:** conceptualization, D.K. and L.C.; methodology, D.K. and L.C.; software, D.K. and L.C.; validation, D.K. and L.C.; formal analysis, D.K. and L.C.; investigation, D.K. and L.C.; resources, D.K. and L.C.; data curation, D.K.; writing—original draft preparation, D.K. and L.C.; writing—review and editing, D.K. and L.C.; visualization, D.K. and L.C.; project administration, D.K. and L.C.; supervision, L.C.; funding acquisition, L.C. All authors have read and agreed to the published version of the manuscript.

**Funding:** This research and APC were co-funded by the Ministry of Science and Higher Education of Poland from Grant 1/S/KPBMiM/20.

**Acknowledgments:** In this section you can acknowledge any support given which is not covered by the author contribution or funding sections. This may include administrative and technical support, or donations in kind (e.g., materials used for experiments).

**Conflicts of Interest:** The authors declare no conflict of interest.

## Appendix A

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```

-----ENCODER-ARDUINO-LAB_VIEW-----
int i = 0;
int gmp = 0;
int check = 0;
void setup()
  { pinMode (2, INPUT);   pinMode (8, OUTPUT);   pinMode (9, OUTPUT);   pinMode (10,
    OUTPUT);   pinMode (11, OUTPUT);   attachInterrupt(0, blinkA, RISING);   Serial.begin (9600);
  }
void loop()
  { while (i != check)
    { check = i;
      if (i % 2 == 0)
        { gmp++; digitalWrite(11, HIGH); digitalWrite(10, HIGH); digitalWrite(9, HIGH);
          digitalWrite(8, HIGH); delay(50);          digitalWrite(11, LOW); digitalWrite(10,
            LOW); digitalWrite(9, LOW);          digitalWrite(8, LOW);          }
        else
          { digitalWrite(10, HIGH);   digitalWrite(8, HIGH);          delay(50);
            digitalWrite(10, LOW);   digitalWrite(8, LOW);          }
        }
    }
void blinkA ()
{i++;}
-----END-----

```

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